

Investigating the transverse motion of a pneumatic shock exciter using two different anvil mounting configurations

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ABSTRACT

This paper describes novel design changes to the accelerometer mounting support of a commercial pneumatic shock exciter, with the aim of reducing the transverse motion the accelerometer is subjected to during shock excitation. The author describes the mounting support supplied by the manufacturer, the design changes made and the measurement data to compare the transfer motions recorded using two different mounting designs.

Section: RESEARCH PAPER

Keywords: transverse motion; shock calibration; shock excitation; accelerometer; pneumatic

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1. INTRODUCTION

Shock measurements [1], [2] have been performed internationally for a long time. In meeting the requirement to provide traceability for the International System of Units (SI), National Metrology Institutes (NMIs) and commercial companies have developed various calibration procedures utilising different shock exciters (SE) [3]-[9]. The operating principal of the SE depends largely on the intended peakacceleration level. For low-level shocks, acceleration levels of $\approx 10 \text{ m/s}^2$, the 'drop ball' or pendulum method is used. For medium-level shocks, acceleration levels in the range of 50 m/s^2 to 10 km/s², a hammer-and-anvil method is employed (rigid body movement of an anvil), using a loaded spring or pneumatic system as the force applicator [10], [11]. For high-shock levels, acceleration levels up to 100 km/s², a Hopkinson bar (shock propagation inside a long, thin bar) is used as the shock exciter [8], [10], [11].

As part of its programme of work, the International Organization for Standardization's Technical Committee (ISO/TC) 108 developed parts 13 and 22 in the ISO 16063 series of standards [10], [11]. Part 13 of the standard specifies primary calibration procedures to determine the shock sensitivity of accelerometers, while part 22 of the standard specifies secondary calibration procedures to determine the shock sensitivity of accelerometers by comparison with a reference transducer.

These procedures have been implemented by various NMIs [3], [5], [6], [9]. The procedures relating to part 13, the implementation thereof and the NMIs' calibration measurement capabilities were validated during the first official international shock comparison, CCAUV.V-K4 [12].

A prominent component in the estimation of the standard uncertainty (UoM) in accelerometer calibrations is the effect of transverse motion (TM) [13]-[16]. As we are dealing with mechanical systems, this component of the UoM cannot be reduced to zero. In practice, the TM uncertainty contribution is a result of two factors:

1) the physical TMs

2) the transverse sensitivity of the accelerometer being calibrated [17]-[21].

As a result, care is taken to reduce the TM of the exciter as much as possible. In the case of the rigid body movement of an anvil system, this is generally achieved by using air-bearing guides.

In this research paper, the author investigates two anvil support configurations implemented in a commercial pneumatic SE system. In Section 2, the system in question is described, outlining the two mounting configurations and the reduced TM envisaged by the implementation of the proposed changes. Section 3 deals with the methods and instrumentation used to perform the TM measurements, which are then reported and discussed in Section 4. Section 3 also describes two different methodologies for considering the results. The author then draws some conclusions in Section 5.

2. ANVIL MOUNTING DESIGN

The accelerometer shock sensitivity calibration system implemented at South Africa's NMI, NMISA, utilises a SPEKTRA SE-201 pneumatic exciter. This exciter uses two anvil systems to cover the complete manufacturer's specified acceleration range from 50 m/s² to 100 km/s². An air-bearing anvil unit is used for the acceleration sub-range from 50 m/s² to 2,500 m/s². For the high-shock (HS) acceleration range, a light-weight aluminium anvil system is used, covering the acceleration range from 2 km/s² to 100 km/s².

For HS, the accelerometer is mounted onto the anvil using a mounting stud. This accelerometer/anvil system forms the rigid body (RB) and is suspended (held in place) using a rubber band (O-ring) in a longitudinal configuration (Mount 1). The O-ring position is shown graphically as the light blue line in Figure 1. The O-ring is hooked around two aluminium pillars mounted on either side of the exciter-system base plate. The pillars have a groove to keep the O-ring in place. This arrangement keeps the RB gently at rest, waiting for the hammer strike. In the present study, we define this axial line as the Y axis.

As can be seen in Figure 1, this arrangement results in two major and opposite tension forces, $T_{S,1.1}$ and $T_{S,1.2}$. There also exist two opposite tension forces, $T_{S,1.3}$ and $T_{S,1.4}$, which are much smaller than, and perpendicular to, $T_{S,1.1}$ and $T_{S,1.2}$. When the dominant forces are in line with the Y axis, this longitudinal configuration will restrain movement along the Y axis, with substantially more freedom of movement (less constraint) along the X axis.

Mount 2, a novel, inexpensive design modification for the mounting configuration, was implemented in this study. The red line in Figure 1. illustrates the configuration for Mount 2. The configuration results in four, almost equal tension forces. Two opposing tension forces, $T_{S,2.1}$ and $T_{S,2.2}$, are perpendicular to $T_{S,2.3}$ and $T_{S,2.4}$. The resulting net force action on the RB results in a more uniform and balanced constraint in both X and Y directions.

The process for modifying the SE high-shock mount from Mount 1 to Mount 2 is very simple and very low cost. The



Figure 2. The new design layout without the triaxial accelerometer for clarity.

supplied O-ring is unbooked from the two suspension pillars and replaced with an O-ring with a larger circumference, $\emptyset \approx 100$ mm. The HS top safety plate is secured using four Allen cap screws and four 20-mm-long spacers with a diameter of 12 mm. The four Allen cap screws are removed one by one, and the O-ring is hooked over the aluminium spacer. Once the O-ring is anchored around all four mounting posts supporting the top plate, the RB is held in position by four almost equal length and perpendicular rubber springs (a section of an O-ring) in an 'X' configuration, as shown in Figure 2. These four sections of the O-ring create almost equal and opposite perpendicular forces, which are applied to the RB at rest and in motion, reducing the resulting TM.

To ensure that the O-ring stays in place (does not move upward because of the shocks), the aluminium spacers were replaced with spacers containing a groove, into which the O-ring was slotted. These spacers (Figure 3) were designed using FreeCAD[©] software and manufactured using 3D-printing technology.



Figure 1. A pictorial representation, showing force vectors, of the two O-ring design layouts investigated. Blue: original design, S1. Red: new design, S2.



Figure 3. Example of a 3D-printed support pillar to replace the aluminium pillar.



Figure 4. Measurement configuration showing the two accelerometers to measure acceleration in X and Y and the laser spot to measure acceleration in Z in the Mount 1 configuration.

3. TRANSVERSE MOTION MEASUREMENTS

The TM (acceleration along the X and Y axes) was measured using a bi-axial accelerometer, while the principal acceleration (Z axis) was measured using laser interferometry. The measurement setup is shown in Figure 4.

3.1. Bi-axial accelerometer

A bi-axial (X–Y) accelerometer was constructed using a 15 x 15 x 15 mm aluminium block. Two Endevco model 7259-100 IEPE accelerometers were stud mounted perpendicular to each other to measure acceleration in the X and Y axes. These accelerometers have a nominal sensitivity of 10 mV/(m/s²) over a wide frequency range, typically 5 Hz to 30 kHz, with a specified peak acceleration of 500 m/s². Even though the TM was to be measured for shock levels of up to 40 km/s², the expected peak TM did not exceed the specified acceleration range of the accelerometer.

The construction of the bi-axial accelerometer was completed with the attachment of an M5 Allen cap screw with a nut on the

opposing side of each accelerometer. These Allen cap screws with nuts were added to allow for a centre of gravity (CG) adjustment to the bi-axial accelerometer. The CG was adjusted to be as close to the centre of the aluminium cube as possible.

3.2. Measurement methodology

Measurement data were collected using a National Instruments PXI unit (NI-PXI). The PXI unit was fitted with three dual channel data acquisition units (DAQ). One DAQ sampled the interferometer I and Q signals used for measuring the shock peak acceleration. The second DAQ was used to sample the output signal of the X- and Y-axis accelerometers, used to measure the transverse acceleration, while the third DAQ was used to trigger (start) each sampling event. The sampling of the three DAQ units was synchronised using the NI niTClk system, which enabled the data sampling to be time synchronised. The calculated average of a set of five measurements was taken as the measurement result for each measurement point (acceleration level).

The resulting transverse acceleration was calculated as

$$a_r = \sqrt{a_X^2 + a_Y^2} , \qquad (1)$$

where a_r is the resulting transverse acceleration, a_X the measured acceleration in the X direction and a_Y the measured acceleration in the Y direction. An example of a measured transverse acceleration is shown in Figure 5. The TM reported and evaluated was the peak transverse acceleration relative to the peak acceleration of interest, the peak acceleration of interest being the peak acceleration along the Z axis. The relative TM was calculated as

$$a_T = \frac{a_r}{a_Z},\tag{2}$$

where a_T is the relative transverse acceleration (RTA), a_r is the resulting TM, calculated using (1), and a_Z is the peak acceleration along the Z axis. Measurements were performed over the acceleration range of 5 km/s² to 40 km/s².

The measurement results revealed a time delay between the reference acceleration (measured by the laser interferometer) and the TM (a_T) peak. The time delay between the two peaks can be



Figure 5. Example of a measured TM during shock measurements, showing a_x , a_y and the resulting a_r .

Table 1. Relative TM mea	asurement results at the	time of the Z-axis peak.
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Table 2. Relative TM measurement results at the time of the RTA peak.

Peak Acceleration	α _τ Mount 1 @ Ref peak	α _τ Mount 2 @ Ref peak	Peak Acceleration	α _τ Mount 1 RTA peak	a _t Mount 2 RTA peak
(km/s²)	(%)	(%)	(km/s²)	(%)	(%)
5	1.9	1.1	5	3.4	1.8
10	1.4	1.2	10	2.6	2.0
20	0.9	0.8	20	1.8	1.2
40	0.6	0.5	40	1.6	1.5

seen in Figure 6. In view of this, two different peak RTA evaluation methodologies were considered and reported on:

- time-delayed peak transverse acceleration (*a_T*), which is the peak TM measured during the complete sampling time,
- 2) transverse acceleration at the time of the reference acceleration peak, which is the transverse acceleration level at the time (instance) of the Z-axis acceleration.

4. MEASUREMENT RESULTS

The RTA was measured at 5, 10, 20 and 40 km/s². Measurements were performed using both mounting support designs described in Section 2, Mount 1 and Mount 2. Furthermore, the RTA was determined at the two instances in time described in Section 3.2. Time instance one at the RTA peak and time instance two synchronised with the Z-axis peak, hence referred to as the @ Ref peak. The results recorded for the two different mounts under investigation, as calculated using @ Ref, are reported in Table 1, with the results recorded for the two different mounts calculated at the RTA peak reported in Table 2.

For accelerometer sensitivity calibration, the metrologist is concerned with the TM severity in the presence of an accelerometer's transverse sensitivity. During rectilinear excitation, it can be assumed that TM is present while the exciter is in motion. Although this is true during the shock excitation described in this research paper, the metrologist is only concerned with the RTA the accelerometer is subjected to during the time/period of the shock pulse.

The data reported in Table 1 reveals that the implementation of Mount 2 marginally improves the RTA compared to Mount 1,

with the RTA improvement being more significant at lower shock acceleration levels. At 5 km/s², the RTA is reduced by about 42 %, which translates into the same percentage reduction in the RTA uncertainty contribution. As the shock level increases from 5 km/s² to 40 km/s², the RTA reduces from 1.9 % to 0.6 % using Mount 1, while the RTA reduces from 1.1 % to 0.5 % using Mount 2. The data reported supports the prediction of reduced TM through the novel anvil suspension design change.

In consideration of the data reported in Table 2, similar trends in the RTA are noted between the two different mountings investigated. The RTA tends to reduce as the shock acceleration level increases. It is noted that the RTA using Mount 2 ('X' configuration) is consistently lower than when using Mount 1, with the largest reduction measured at 5 km/s². However, it is necessary to consider the data carefully within the conditions of the data assessment.

As pointed out earlier, as the shock level increases, the time difference between the Z-axis shock peak and the RTA peak increases; that is, the RTA peak is reached some time after the principal axis acceleration peak. For shocks with peak acceleration levels above 10 km/s², the RTA peak occurred after the end of the sampling time. The aim of the investigation was not to determine the actual peak TM, but the TM that influences the uncertainty in shock sensitivity calibration.

5. CONCLUSIONS

The performance with respect to the resulting TM of two different mounting configurations used to mount/support the anvil of a pneumatic SE were investigated. Two different instances of peak TM were also considered, first, at the instance



Figure 6. Shock (left vertical axis) and transverse acceleration (right vertical axis) time series.

in time when the peak acceleration occurs, and second, when the resulting TM (a_r) peak occurs, which is generally some time after the principal axis acceleration peak. In terms of the influence of RTA as an uncertainty contributor, the author focused on the results from the first instance.

The results indicate that the configuration in Mount 2 provides a reduction in TM of almost 50 % at shocks of 5 km/s². The performance gain using Mount 2 reduces as the shock acceleration increases to almost 0 for peak shock levels above 40 km/s^2 .

The benefit (with possible reduced expanded uncertainty of measurement) of implementing the proposed mounting configuration is worth the minimal mechanical changes required to implement it. It should be noted that with this mounting configuration, mounting the accelerometer onto the anvil is a little more time consuming than the manufacturer's original configuration.

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